

CMPE-242

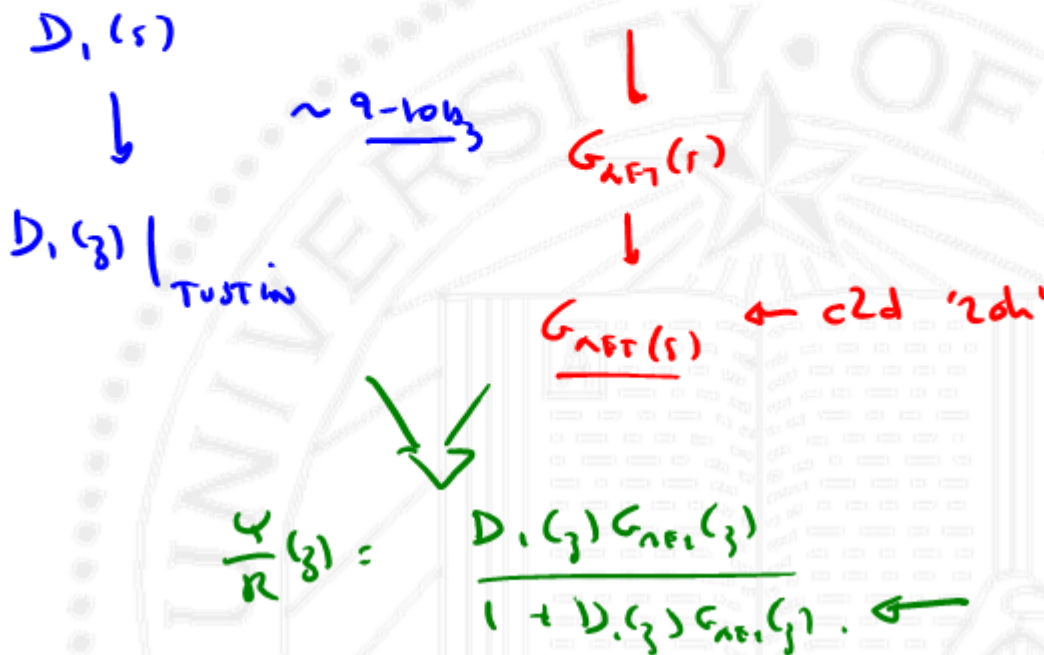
Applied Feedback Control

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Converting things to discrete w/ different b/c's.

Stability should be preserved.



STATE SPACE

Control \rightarrow LQR

ESTIMATOR \rightarrow LQE / KALMAN FILTER

STATE COMMAND
| BWS ESTIMATION

ABUSE LQR TO DO FUN THINGS



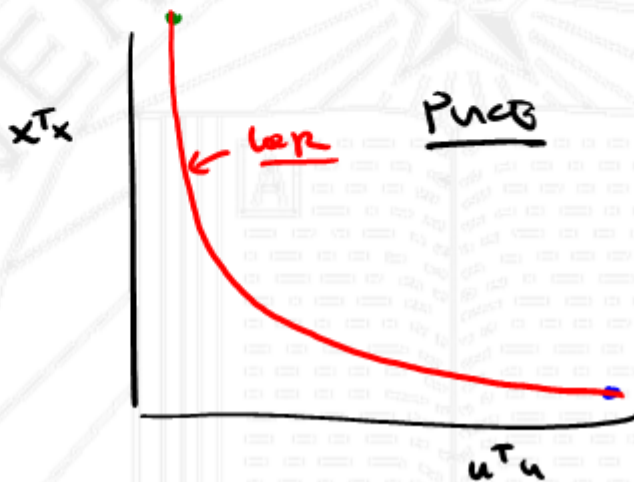
regulation

↓
LQR:

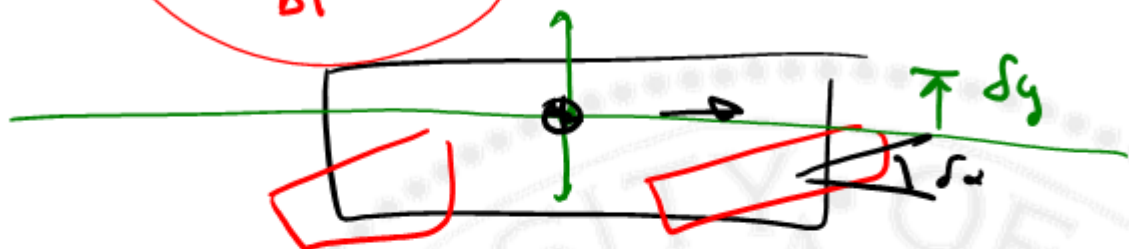
$$\min_u \int_0^{\infty} (x^T Q x + u^T R u) dt$$

$$\text{subj: } \dot{x} = Ax + Bu$$

Solution: $u = -Kx$ ← optimum solution



$$\delta \dot{y} \approx \frac{\delta y_k - \delta y_{k-1}}{\delta T}$$

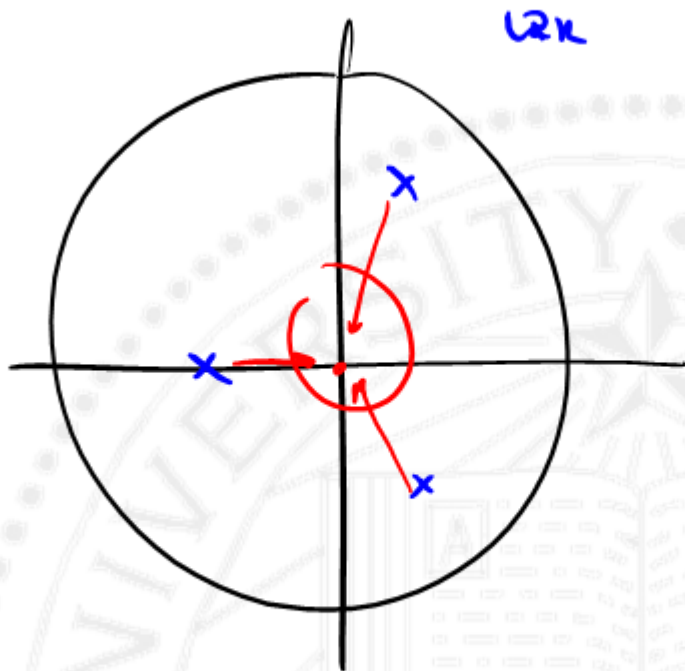


$$x = \begin{bmatrix} \delta y \\ \delta y \\ \delta x \end{bmatrix}$$

$$\dot{x} = \begin{bmatrix} \delta \dot{y} \\ \delta \dot{y} \\ \delta \dot{x} \end{bmatrix}$$

$$C = \begin{bmatrix} 0 & 1 & 0 \\ 0 & 0 & 1 \end{bmatrix}$$





$\omega_{2\pi}$

Poles
5-10.



Kalman Filter

$$\dot{x} = Ax + Bu + \Gamma w$$

process noise

$$y = Cx + d$$

measurement noise

$$w \sim N(0, R_w)$$
$$d \sim N(0, R_d)$$

$R_w \gg R_r$ - believe sensors
 $L \leftarrow$ big

$R_w \ll R_r$ - believe model
 $L \leftarrow$ small.

$$\dot{\hat{x}} = A\hat{x} + Bu - L(y - \hat{y})$$

Kalman Filter design "L"

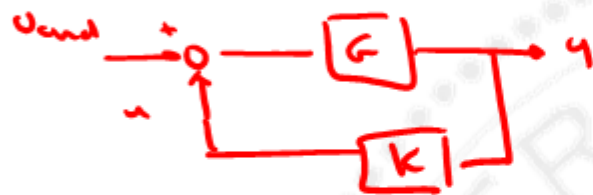
$L = \text{lqr}(A, \Gamma, C, R_w, R_r)$ - Kalman Filter (study sheet)

$$L^T = \text{lqr}(A^T, C^T, \underbrace{\Gamma R_w \Gamma^T}_a, \underbrace{R_r}_r)$$





output follows input
 "AIRBUS"
 FUN AUTOPILOT



Augment stability of the plant
 "747"
 STABILITY AUGMENTATION



DIGITAL

$$\dot{x} = Ax + Bu$$

$$y = Cx + Du$$



$$x_{k+1} = \Phi x_k + \Gamma u_k$$

$$y_k = Hx_k + Du_k$$

$$\Phi = e^{A\Delta T}$$

← matrix exponential: $e^{At} = I + At + \frac{A^2 t^2}{2!} + \dots$

$$\Gamma = \left[\int_0^{\Delta T} e^{A\tau} d\tau \right] B$$

$$H = C \quad D = D$$

DISCRETE EQUIVALENTS TO

$$\dot{x} = Ax + Bu$$

$$y = Cx + Du$$

z domain → ' z domain'



$$\text{lqr} - J = \int_0^{\infty} x^T Q x + u^T R u \, dt$$

$$\text{dlqr} - J = \sum_{k=0}^{\infty} x_k^T Q x_k + u_k^T R u_k$$

lqr overloaded

$$K = \text{lqr}(A, B, Q, R)$$

$$K = \text{lqr}(\Phi, \Gamma, Q, R)$$

$$u_k = -K(\hat{x}_k - \underline{x}_e)$$

↙ control state.



$$x_{k+1} = \Phi x_k + \Gamma u_k$$

$$u_k = -K(\hat{x}_k - x_k^e)$$

$$y_k = Hx_k + Du_k$$

$$\begin{cases} \hat{x}_{k+1} = \Phi \hat{x}_k + \Gamma u_k + L(y_k - \hat{y}_k) = [\Phi - L H] \hat{x}_k + [\Gamma \ L] \begin{bmatrix} u_k \\ y_k \end{bmatrix} \\ \hat{y}_k = H \hat{x}_k + D u_k \end{cases}$$

initialize \hat{x}_k, u_k



measure y_k

$$\hat{x}_{k+1} = [\Phi - L H] \hat{x}_k + \Gamma u_k + L y_k$$

$$u_{k+1} = -K(\hat{x}_{k+1} - x_{k+1}^e)$$

$$\hat{x}_k \leftarrow \hat{x}_{k+1}$$

$$u_k \leftarrow u_{k+1}$$



Problems w/ Kalman

Time varying are numerically unstable.

Square root form of KF.

Information form of KF

$$S = \bar{P}^{-1}$$

$$P = E(\hat{x} \hat{x}^T)$$

process noise } ergodic
measurement noise }

STAT'S DON'T CHANGE

E.K.F

↑

$$\dot{x} = f(x, u)$$

at each \hat{x}_k

$$A = \left. \frac{\partial f}{\partial x} \right|_{\hat{x}_k, \hat{u}_k}$$

$$B = \left. \frac{\partial f}{\partial u} \right|_{\hat{x}_k, \hat{u}_k}$$

"white"
|
(0, σ^2)



Continuous Time

$$\dot{x} = Ax + Bu$$

$$y = Cx + Du$$

$$u = -kx \quad / \quad -k\hat{x}$$

$$\dot{\hat{x}} = A\hat{x} + Bu + L(y - \hat{y})$$

$$\hat{y} = C\hat{x} + Du$$

$$K = \text{place}(A, B, P_{cl})$$

$$L^T = \text{place}(A^T, C^T, P_{obs})$$

$$K = \text{lqr}(A, B, Q, R)$$

$$L = \text{lqe}(A, B, R_w, R_d)$$

$$\Phi = e^{A\Delta T}$$
$$P = \int_0^{\Delta T} e^{A\tau} d\tau \cdot B$$

Discrete

$$x_{k+1} = \Phi x_k + \Gamma u_k$$

$$y_k = Hx_k + Du_k$$

$$u_k = -kx_k \quad / \quad -k\hat{x}_k$$

$$\hat{x}_{k+1} = \Phi\hat{x}_k + \Gamma u_k + L(y_k - \hat{y}_k)$$

$$\hat{y}_k = H\hat{x}_k + Du_k$$

$$K = \text{place}(\Phi, \Gamma, P_{cl})$$

$$L^T = \text{place}(\Phi^T, H^T, P_{obs})$$

$$K = \text{dlqr}(\Phi, \Gamma, Q, R)$$

$$L = \text{dlqe}(\Phi, H, R_w, R_d)$$



$\phi = e^{AT}$ ← matrix exponential (expm)

$$e^{At} \triangleq I + At + \frac{A^2 t^2}{2!} + \dots$$

$$\Gamma = \int_0^{bT} e^{A\tau} d\tau \cdot B \quad \text{— zoh equivalent for input}$$

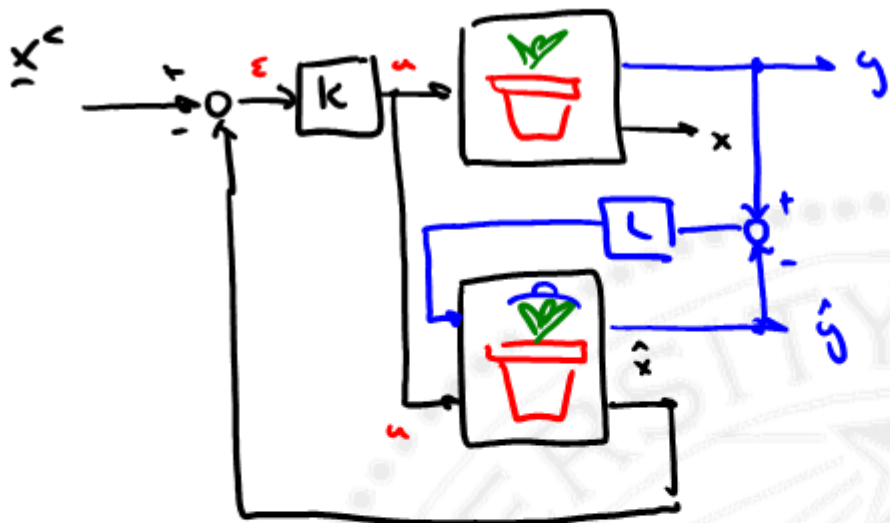
$$\dot{x} = Ax \rightarrow x(t) = e^{At} x(0)$$

c2d 'zoh'

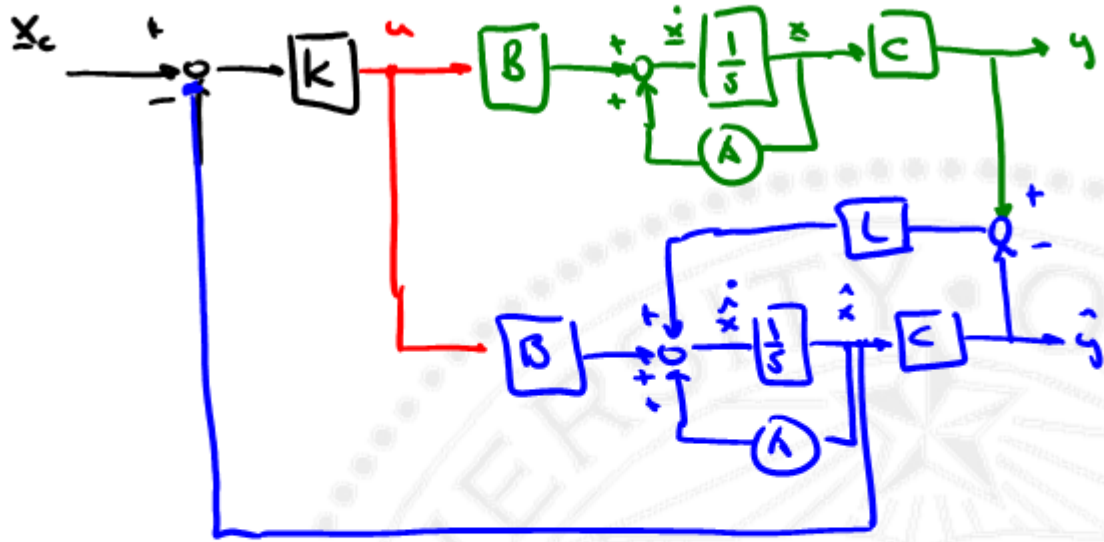
$R_w \neq R_{wk}$

← disrw ()

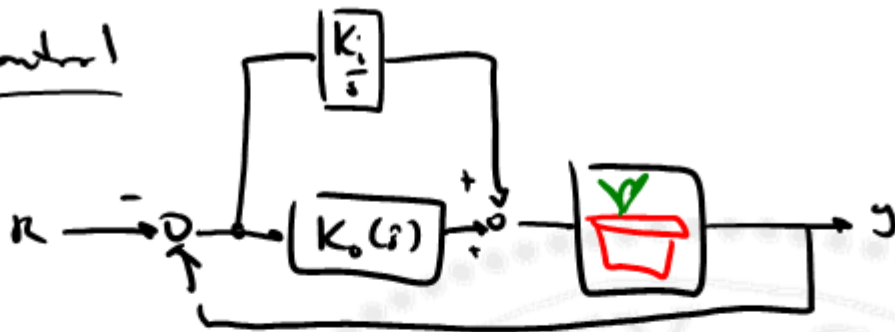




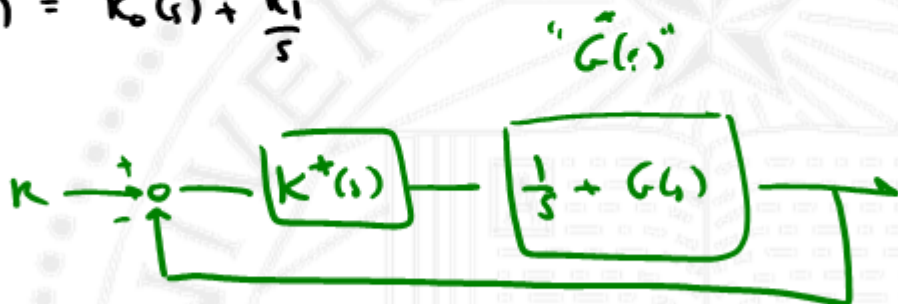
$$x_{k+1} = \frac{1}{3} x_k$$



Integral Control



$$K^*(s) = K_o(s) + \frac{K_i}{s}$$



$$\dot{x} = Ax + Bu$$

$$y = Cx$$

$$e = r - y = r - Cx$$

$$x_I = \int e dt \therefore \dot{x}_I = e = r - Cx$$

$$\begin{bmatrix} \dot{x} \\ \dot{x}_I \end{bmatrix} = \underbrace{\begin{bmatrix} A & \phi \\ -C & \rho \end{bmatrix}}_{A^{aug}} \begin{bmatrix} x \\ x_I \end{bmatrix} + \underbrace{\begin{bmatrix} B \\ 0 \end{bmatrix}}_{B^{aug}} u + \begin{bmatrix} 0 \\ \dots \\ I \end{bmatrix} r$$

$$u = - \begin{bmatrix} k & k_I \end{bmatrix} \begin{bmatrix} x \\ x_I \end{bmatrix}$$

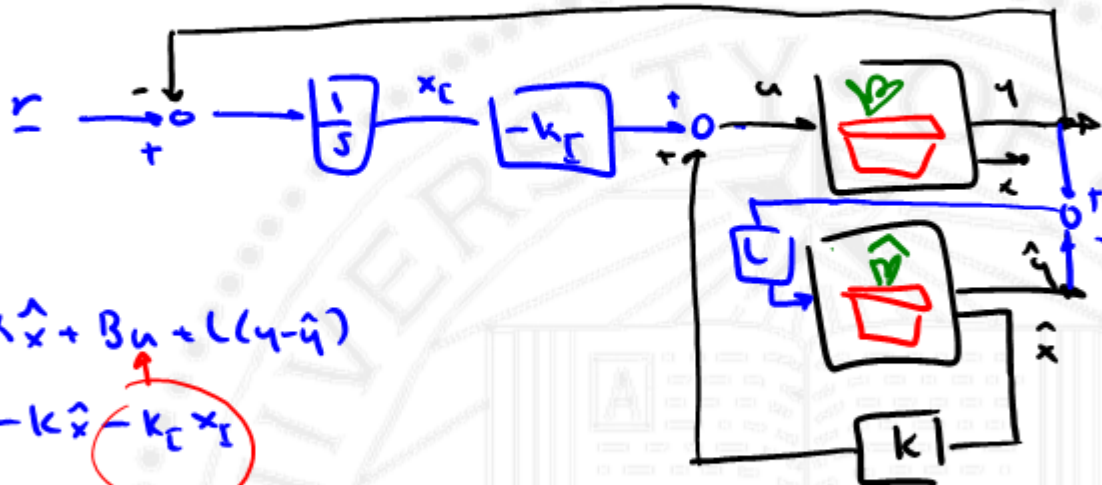
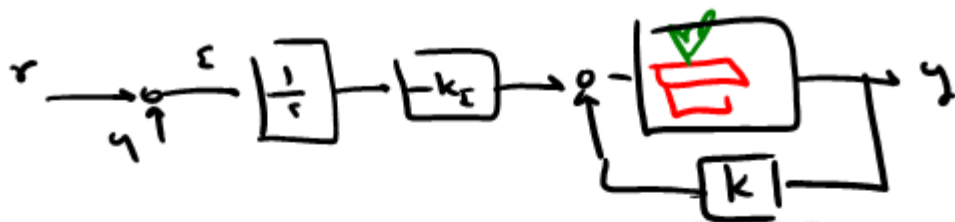
$$\text{eig}(A^{aug} - B^{aug}(k \cdot k_I))$$



$$x \in \mathbb{R}^5 \quad u \in \mathbb{R}^2$$

- Drive as many states to a reference as I have actuators/inputs
- Drive under a linear combination of states to reference as I have actuators/inputs





$$\dot{\hat{x}} = A\hat{x} + Bu + L(y - \hat{y})$$

$$u = -k\hat{x} - k_I x_I$$



$$\begin{bmatrix} \dot{x} \\ \vdots \\ \dot{x}_r \end{bmatrix} = \underbrace{\begin{bmatrix} A & 0 \\ -c & 0 \end{bmatrix}}_{A_n} \underbrace{\begin{bmatrix} x \\ \vdots \\ x_r \end{bmatrix}}_{x_n} + \underbrace{\begin{bmatrix} B & 0 \\ 0 & I \end{bmatrix}}_{B_n} \underbrace{\begin{bmatrix} u \\ r \end{bmatrix}}_{u_n}$$

$$\dot{x}_n = A_n x_n + B_n u_n$$

$$y_n = C_n x_n$$

$$\dot{\hat{x}}_n = A_n \hat{x}_n + B_n u_n + L(y_n - \hat{y}_n)$$

$$\hat{y}_n = C_n \hat{x}_n$$

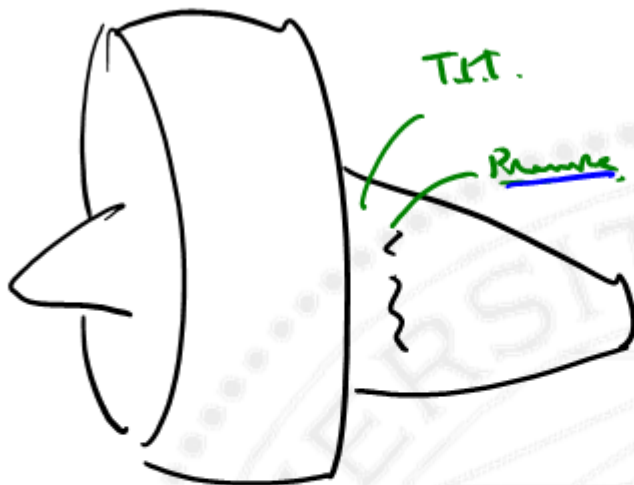
$$u_n = -k x_n = -k_c \hat{x} - k_I \hat{x}_I$$

reduced order estimator

k_c



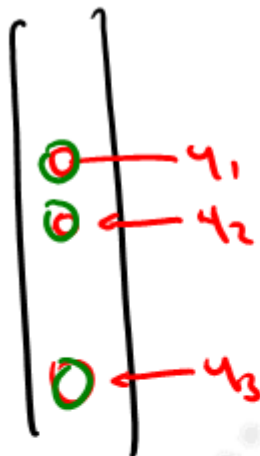
REDUCED ORDER ESTIMATOR



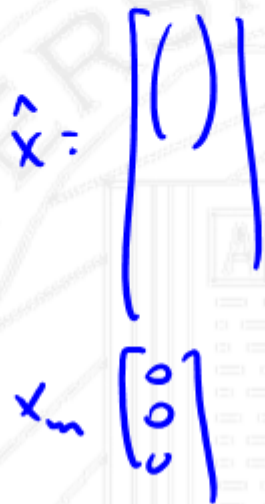
$$u = -k\hat{x}$$



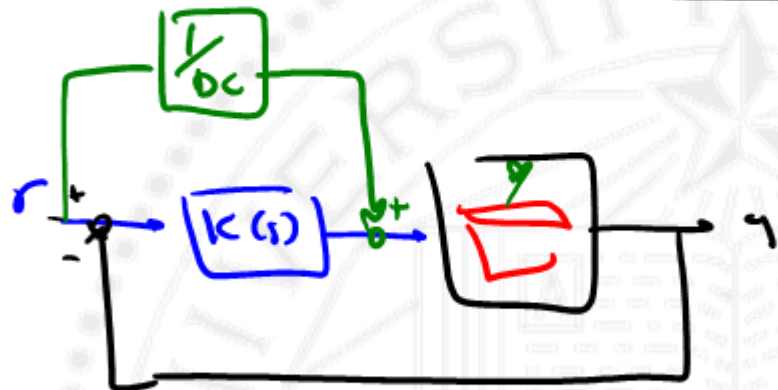
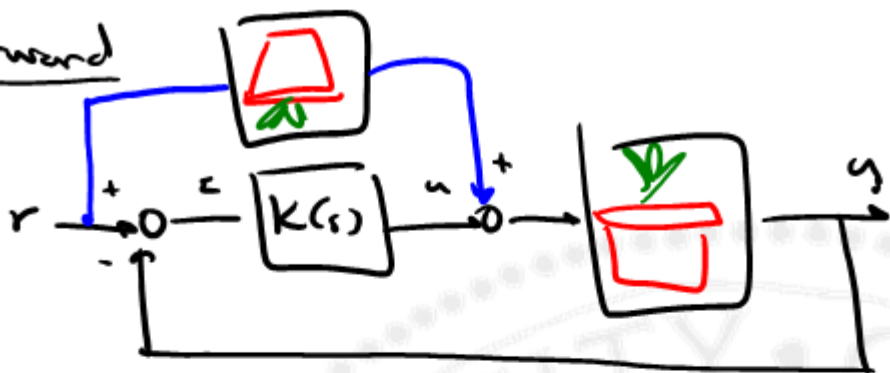
\hat{x}

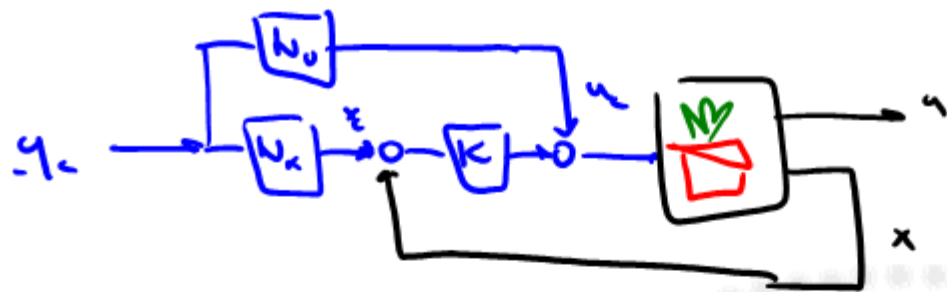


$$u = -k \hat{x}$$



Feed Forward





N_x converts $y_c \rightarrow x_c$

N_u converts $y_c \rightarrow y_{ss}$

FPE 7.10

